

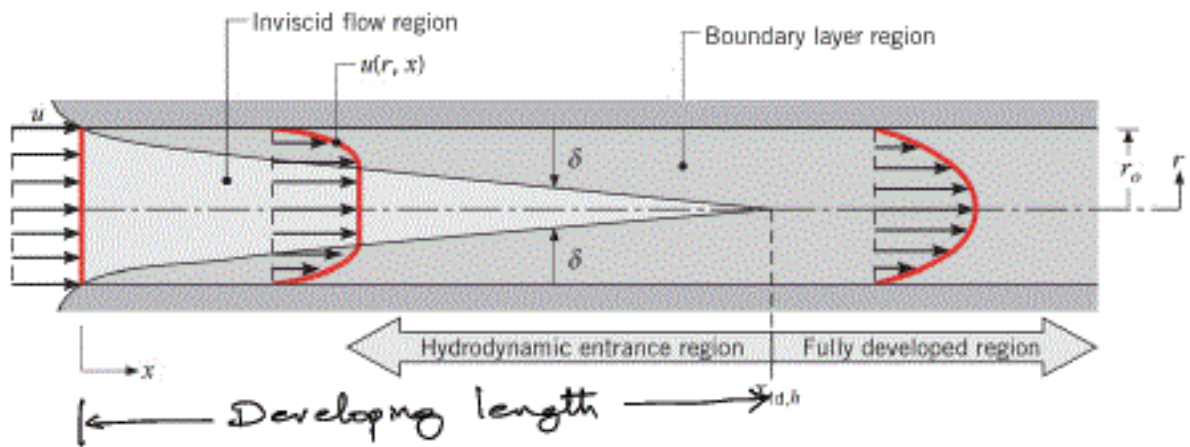


What we will be looking at;

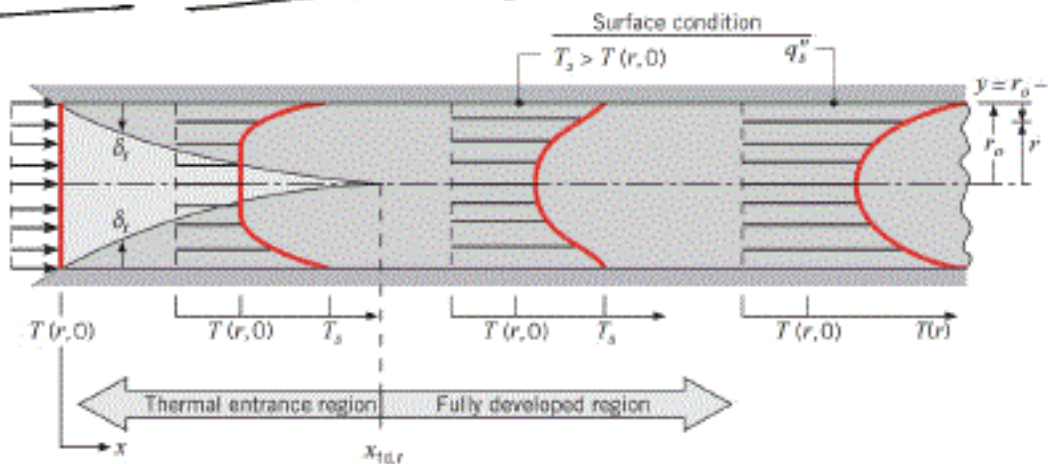
- ① No phase change [Single phase]
- ② Governing equations [Correlations]

Internal flow:

Hydrodynamic Boundary layer

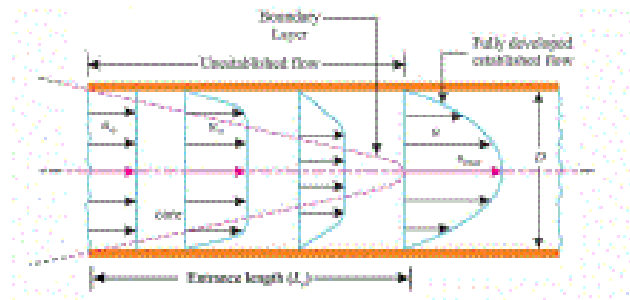
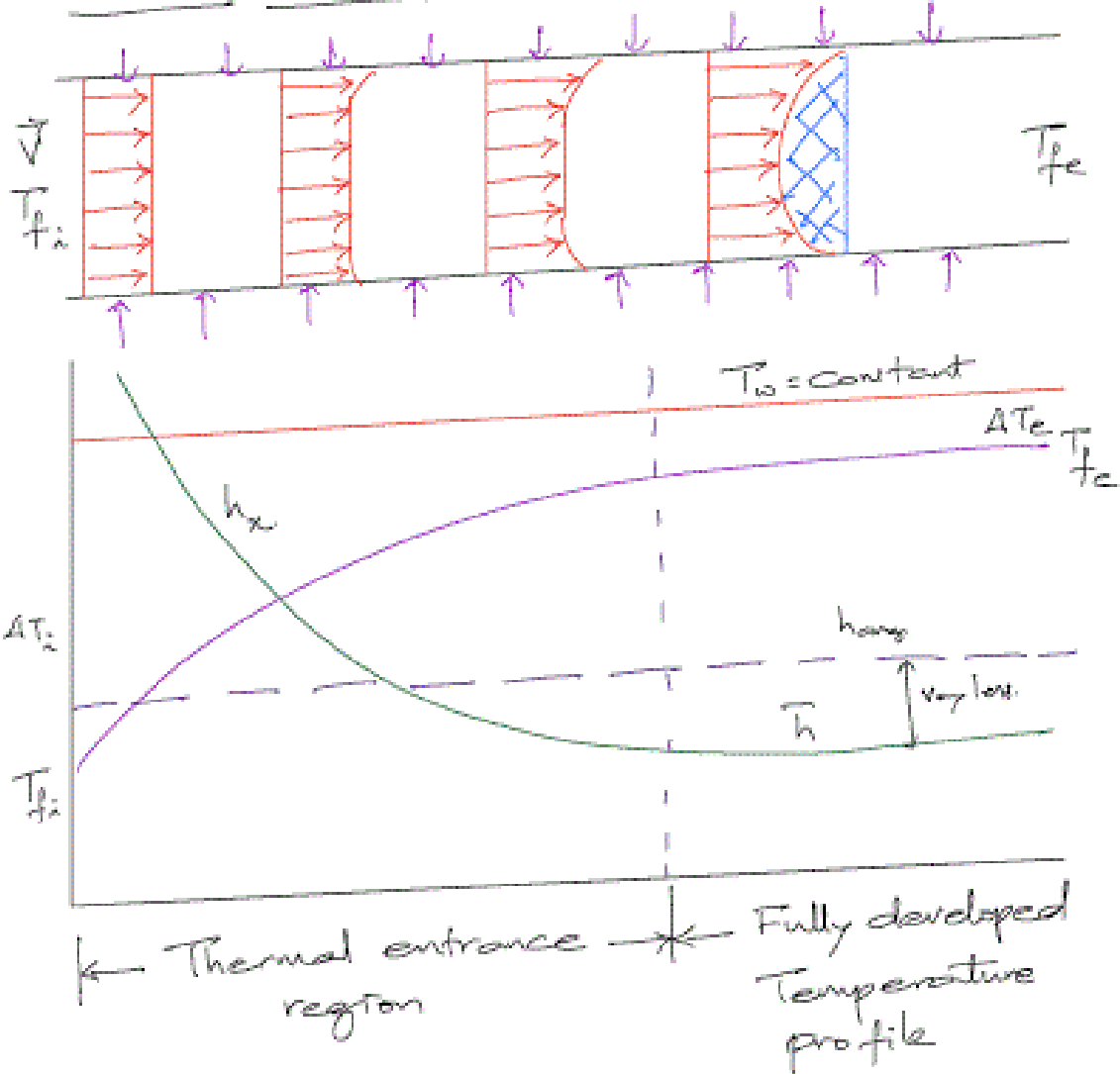


Thermal boundary layer:



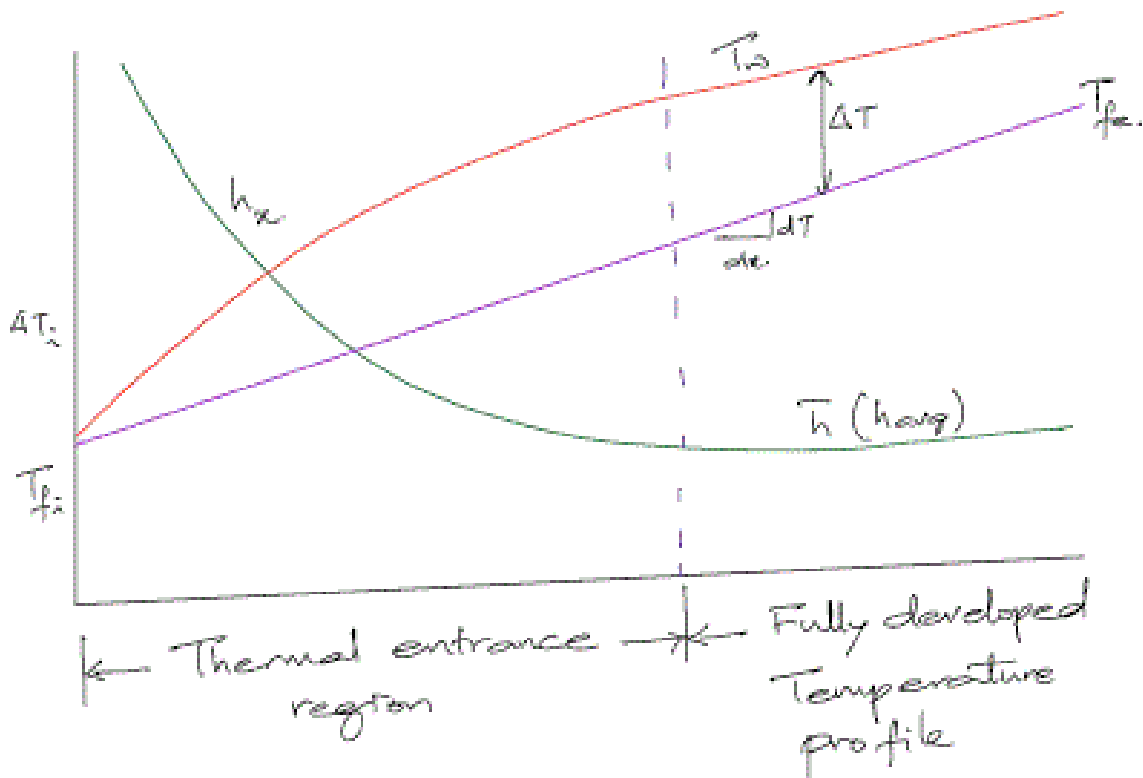
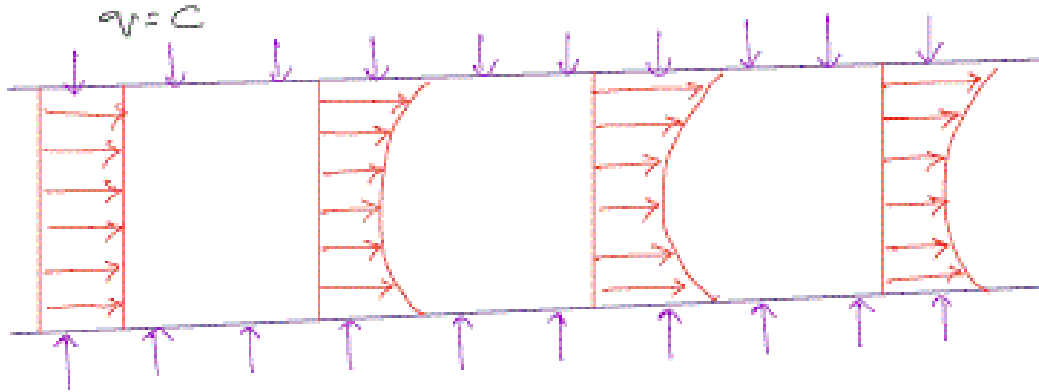


Constant wall temperature ($T_w = \text{constant}$)





Constant wall heat flux [$q_w = \text{constant}$] $\frac{dT}{dx} (\text{linear})$





Things which are important for such flows.

1) Average velocity $\bar{u} = \frac{1}{A} \int u_z dA$.

2) Mass flow rate $\dot{m} = \rho A \bar{u}$; $Q = A \bar{u}$; $A = \frac{\pi d^2}{4}$.

3) Bulk mean temperature: It is the temperature at a location 'x' along the pipe, which is the average temperature of the fluid such that the fluid at that location is well mixed to attain a common temperature. Mathematically,

$$T_b = \frac{1}{\bar{u} A} \int u T dA \quad T_b = \frac{T_{fi} + T_{fc}}{2}$$

4) Friction factor [f]

Laminar $\rightarrow f = \frac{64}{Re}$

Turbulent \rightarrow Refer to Moody's diagram.

5) Pressure drop $\Delta P = f \times \frac{\rho \bar{u}^2}{2} \times \frac{L}{d}$

6) Reynolds number Re

$Re < 2300 \rightarrow$ Laminar.

$Re > 2300 \rightarrow$ Turbulent.

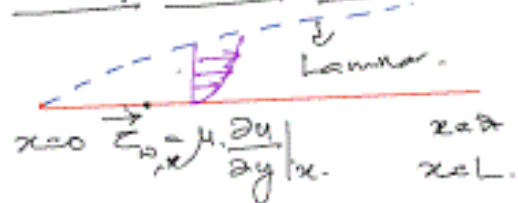


Flow over flat plates:

Drag coefficient (or) skin friction coefficient

i) Local drag coefficient:

$$C_{f,x} = \frac{\tau_{w,x}}{\frac{\rho u_{\infty}^2}{2}}$$



ii) Average drag coefficient

$$C_{f,avg} = \frac{1}{L} \int_0^L C_{f,x} dx$$

$$\tau_{w,avg} = \frac{\text{Drag force}}{\text{Area}}$$

$$\tau_{w,avg} = \frac{F_D}{A}$$

$$C_{f,avg} = \frac{\tau_{w,avg}}{\frac{1}{2} \rho u_{\infty}^2}$$

$$\left. \begin{aligned} Re_x &= \frac{\rho V x}{\mu} \rightarrow \text{Local} \\ Re_L &= \frac{\rho V L}{\mu} \rightarrow \text{Average} \end{aligned} \right\}$$

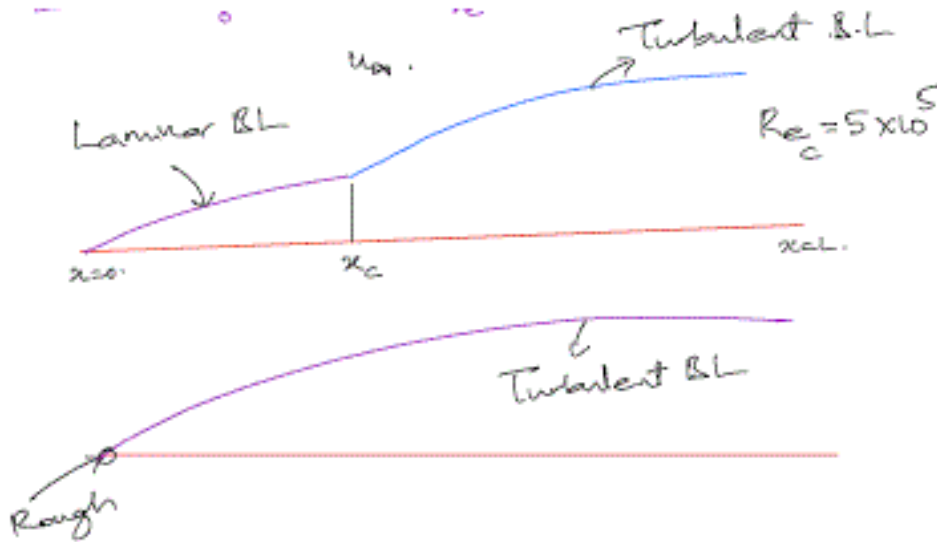
Drag on a plate:

$$\left. \begin{aligned} C_{f,x} &= 0.664 Re_x^{-0.5} \\ C_{f,L} &= 1.328 Re_L^{-0.5} \end{aligned} \right\} \text{for } Re < 10^5, \text{ Laminar}$$

$$\left. \begin{aligned} C_{f,x} &= 0.0592 Re_x^{-0.2} \\ C_{f,x} &= 0.37 [\log_{10} Re_x]^{-2.584} \end{aligned} \right\} \begin{aligned} 5 \times 10^5 &< Re < 10^7 \\ Re &> 10^7 \end{aligned} \text{ Turbulent}$$

$$C_{f,L} = 0.074 Re_L^{-0.2} - 1742 Re_L^{-1} \rightarrow \text{Laminar \& Turbulent.}$$

$$[C_{f,L} = \frac{1}{L} \int_0^{x_0} C_{f,x, \text{laminar}} dx + \int_{x_0}^L C_{f,x, \text{turbulent}} dx]$$

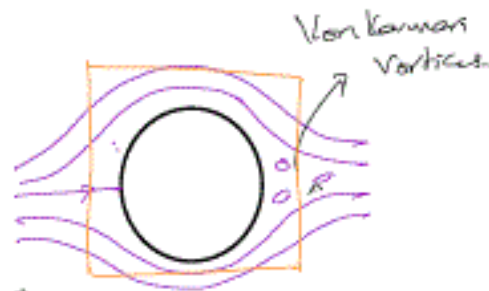


Flow over cylinder

$$A_s = \pi DL$$

$$Re_D = \frac{\rho V D}{\mu} = \frac{V D}{\nu}$$

$$C_D = \frac{F_D/A}{\frac{1}{2} \rho u_m^2} \rightarrow \text{Projected Area.}$$



Correlation from experimental data

$$C_D = 10.41 Re_D^{-0.67} \rightarrow 0.1 < Re_D < 4$$

$$C_D = 5.67 Re_D^{-0.25} \rightarrow 4 < Re_D < 1000$$

$$C_D = 1 \rightarrow 1000 < Re_D < 5000$$

$$C_D = 0.310 Re_D^{0.1525} \rightarrow 5000 < Re_D < 10^4$$

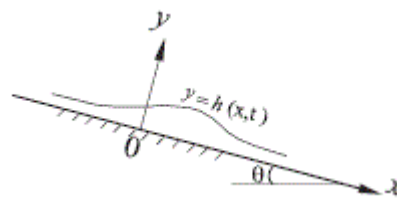
$$C_D = 1.14 \rightarrow 10^4 < Re_D < 2 \times 10^5$$



Inclined flow introduction:

Because of the inherent nonlinearity, most fluid-dynamical problems must be solved by either analytical approximations or by numerical computations. In this chapter we shall first explain the lubrication approximation, with applications to the flow of thin layers. Attention will then be turned to the slow flow past a sphere and a cylinder. Application to the environmental problem of aerosols will be briefly discussed. Lastly the slow withdrawal of fluid from a stratified reservoir into a sink will be analyzed.

consider the flow of a thin layer of viscous fluid on an inclined plane. Referring to Figure let the x - axis coincide with the plane bed inclined at the angle θ with respect to the horizon, and the y - axis normal to the plane bed,



A thin fluid layer flowing down an incline plane