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DEPARTMENT OF AEROSPACE ENGINEERING

19ASB304 - COMPUTATIONAL FLUID DYNAMICS FOR AEROSPACE APPLICATIONS III YEAR VI SEM

UNIT-II FINITE ELEMENT TECHNIQUES
TOPIC: Stability properties of explicit and implicit methods

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Explicit Method

$$\frac{u_i^{n+1} - u_i^n}{\Delta t} = \alpha \left[\frac{u_{i+1}^n - 2u_i^n + u_{i-1}^n}{\Delta x^2} \right]$$

- Explicit method uses the fact that we know the dependent variable, u at all x at time t from initial conditions
- Since the equation contains only one unknown, u_iⁿ⁺¹ (i.e. u at time t+∆t), it can be obtained directly from known values of u at t
- The solution takes the form of a "marching" procedure in steps of time

Crank - Nicolson Implicit Method

- The unknown value u at time level (n+1) is expressed both in terms of known quantities at n and unknown quantities at (n+1).
- The spatial differences on RHS are expressed in terms of averages between time level n and (n+1):

$$\frac{u_i^{n+1} - u_i^n}{\Delta t} = \frac{\alpha}{2} \left[\frac{u_{i+1}^{n+1} + u_{i+1}^n - 2u_i^{n+1} - 2u_i^n + u_{i-1}^{n+1} + u_{i-1}^n}{\Delta x^2} \right]$$

- The above equation cannot result in a solution of u_iⁿ⁺¹ at grid point i.
- The eq. is written at all grid points resulting in a system of algebraic equations which can be solved simultaneously for u at all i at time level (n+1).

Crank – Nicolson Implicit Method

The equation can be rearranged as

$$-u_{i-1}^{n+1} + \frac{2+2r}{r}u_i^{n+1} - u_{i+1}^{n+1} = u_{i-1}^n + \frac{2-2r}{r}u_i^n + u_{i+1}^n$$
 where $r = \alpha \Delta t/(\Delta x)^2$

On application of eq. at all grid points from i=1 to i=k+1, the system of eqs. with boundary conditions u=A at x=0 and u=D at x=L can be expressed in the form of Ax = C

$$\begin{bmatrix} B(1) & -1 & 0 & 0 & \dots & 0 \\ -1 & B(2) & -1 & 0 & \dots & 0 \\ 0 & -1 & B(3) & -1 & \dots & 0 \\ \vdots & & & & & \\ 0 & 0 & 0 & \dots & -1 & B(k-1) \end{bmatrix} \begin{bmatrix} u_2^{n+1} \\ u_3^{n+1} \\ u_4^{n+1} \\ \vdots \\ u_k^{n+1} \end{bmatrix} = \begin{bmatrix} (C(1)+A)^n \\ C(2)^n \\ C(3)^n \\ \vdots \\ (C(k-1)+D)^n \end{bmatrix}$$

 A is the tridiagonal coefficient matrix and x is the solution vector. The eq. can be solved using Thomas Algorithm



Explicit ~ Implicit – A Comparison

Explicit Method

- Easy to set up.
- Constraint on mesh width, time-step.
- Less computer time.

Implicit Method

- Complicated to set up.
- Larger computer time.
- No constraint on time step.
- Can be solved using Thomas Algorithm.

Consistency

A finite difference representation of a PDE is said to be consistent if:

$$\lim_{mesh\to 0} (PDE - FDE) = \lim_{mesh\to 0} (TE) = 0$$

- For equations where truncation error is $O(\Delta x)$ or $O(\Delta t)$ or higher orders, TE vanishes as the mesh is refined
- However, for schemes where TE is $\mathcal{O}(\Delta t/\Delta x)$, the scheme is not consistent unless mesh is refined in a manner such that $\Delta t/\Delta x \rightarrow 0$
- For the Dufort-Frankel differencing scheme (1953), if Δt/Δx does not tend to zero, a parabolic PDE may end up as a hyperbolic equation

Convergence

A solution of the algebraic equations that approximate a PDE is convergent if the approximate solution approaches the exact solution of the PDE for each value of the independent variable as the grid spacing tends to zero:

$$u_i^n = \bar{u}(x_i, t_n) \text{ as } \Delta x, \Delta t \to 0$$

RHS is the solution of algebraic equation

Errors & Stability Analysis

Errors:

- A = Analytical solution of PDE
- D = Exact solution of finite difference equation
- N = Numerical solution from a real computer with finite accuracy
- Discretization Error = A -D = Truncation error + error introduced due to the treatment of boundary condition
- Round-off Error = ε= N –D

$$N = \varepsilon + D$$

ε will be referred to as "error" henceforth

Errors & Stability Analysis

Consider the I-D unsteady state heat conduction equation and its FDE:

$$\frac{\partial u}{\partial t} = \alpha \frac{\partial^2 u}{\partial x^2} \implies \frac{u_i^{n+1} - u_i^n}{\Delta t} = \alpha \left[\frac{u_{i+1}^n - 2u_i^n + u_{i-1}^n}{\Delta x^2} \right]$$

N must satisfy the finite difference equation :

$$\frac{D_i^{n+1} + \varepsilon_i^{n+1} - D_i^n - \varepsilon_i^n}{\Delta t} = \alpha \left[\frac{D_{i+1}^n + \varepsilon_{i+1}^n - 2D_i^n - 2\varepsilon_i^n + D_{i-1}^n + \varepsilon_{i-1}^n}{\Delta x^2} \right]$$

Also, D being the exact solution also satisfies FDE:

$$\frac{D_i^{n+1} - D_i^n}{\Delta t} = \alpha \left[\frac{D_{i+1}^n - 2D_i^n + D_{i-1}^n}{\Delta x^2} \right]$$

Subtracting above 2 equations, we see that error ε also satisfies FDE :

$$\frac{\varepsilon_i^{n+1} - \varepsilon_i^n}{\Delta t} = \alpha \left[\frac{\varepsilon_{i+1}^n - 2\varepsilon_i^n + \varepsilon_{i-1}^n}{\Delta x^2} \right]$$

If errors ε_i's shrink or remain same from step n to n+1, solution is stable.
Condition for stability is:

$$\left|\frac{\varepsilon_i^{n+1}}{\varepsilon_i^n}\right| \le 1$$

Application in Fluid Flow Equations

Introduction

- Fluid mechanics: More complex, governing PDE's form a nonlinear system.
- nonlinear system.

 Burger's Equation: $\frac{\partial \zeta}{\partial t} + u \frac{\partial \zeta}{\partial x} = v \frac{\partial^2 \zeta}{\partial x^2} = > Includes time dependent, convective and diffusive term.$
- Here 'u': velocity, 'γ': coefficient of viscosity, & 'ζ': any property which can be transported or diffused.
- Neglecting viscous term, remaining equation is a simple analog of Euler's equation : $\frac{\partial \zeta}{\partial t} + u \frac{\partial \zeta}{\partial r} = 0$

Conservative Property

- FDE possesses conservative property if it preserves integral conservation relations of the continuum
- Consider Vorticity Transport Equation: $\frac{\partial \omega}{\partial t} = -(V \cdot \nabla)\omega + \nu \nabla^2 \omega$

where ∇ is nabla, V is fluid velocity and ω is vorticity.

Integrating over a fixed region we get,

$$\int_{\Re} \frac{\partial \omega}{\partial t} d\Re = -\int_{\Re} (V \cdot \nabla) \omega d\Re + \int_{\Re} \nu \nabla^2 \omega d\Re$$

which can be written as:

$$\frac{\partial}{\partial t} \int_{\Re} \omega \, d\Re = - \int_{A_o} (V\omega) \cdot n \, dA + \nu \int_{A_o} (\nabla \omega) \cdot n \, dA$$

i.e. rate of accumulation of ω in \Re is equal to net advective flux rate plus net diffusive flux rate of ω across A_o into \Re

The concept of conservative property is to maintain this integral relation in finite difference representation.

Conservative Property

Consider inviscid Burger's equation :

$$\frac{\partial \omega}{\partial t} = -\frac{\partial}{\partial x}(u\omega) \quad \text{FDE Analog} \qquad \frac{\omega_i^{n+1} - \omega_i^n}{\Delta t} = -\frac{u_{i+1}^n \omega_{i+1}^n - u_{i-1}^n \omega_{i-1}^n}{2\Delta x}$$

Evaluating the integral $\frac{1}{\Delta t} \sum_{i=l_1}^{l_2} \omega \Delta x$ over a region running from i=I₁ to i=I₂: $\frac{1}{\Delta t} \left[\sum_{i=l_1}^{l_2} \omega_i^{n+1} \Delta x - \sum_{i=l_2}^{l_2} \omega_i^{n} \Delta x \right] = (u\omega)_{l_1 - \frac{1}{2}} - (u\omega)_{l_2 + \frac{1}{2}}$

$$\frac{1}{\Delta t} \left[\sum_{i=l_1}^{l_2} \omega_i^{n+1} \Delta x - \sum_{i=l_1}^{l_2} \omega_i^n \Delta x \right] = (u\omega)_{l_1 - \frac{1}{2}} - (u\omega)_{l_2 + \frac{1}{2}}$$

Thus, the FDE analogous to inviscid part of the integral has preserved the conservative property.

For non-conservative form of inviscid Burger's equation: $\frac{\partial \omega}{\partial t} = -u \frac{\partial \omega}{\partial x}$

$$\frac{1}{\Delta t} \left[\sum_{i=l_1}^{l_2} \omega_i^{n+1} \Delta x - \sum_{i=l_1}^{l_2} \omega_i^n \Delta x \right] = \frac{1}{2} \sum_{i=l_1}^{l_2} \left[u_i^n \omega_{i-1}^n - u_i^n \omega_{i+1}^n \right]$$

i.e. FDE analog has failed to preserve the conservative property

Transportive Property

- FDE formulation of a flow is said to possess the transportive property if the effect of perturbation is convected only in the direction of velocity
- Consider model Burger's equation in conservative form and a perturbation $ε_m = δ$ in ζ for u>0, all other ε=0
- Using FTCS, we find the transportive property to be violated
- On the contrary when an upwind scheme is used,

$$\frac{\zeta_i^{n+1} - \zeta_i^n}{\Delta t} = -\frac{u\zeta_i^n - u\zeta_{i-1}^n}{\Delta x}$$

$$= \sum_{m+1}^{m+1} - \zeta_{m+1}^n = +\frac{u\delta}{\Delta x}$$

$$= \sum_{m+1}^{m+1} - \zeta_{m+1}^n = -\frac{u\delta}{\Delta x}$$

$$= \sum_{m+1}^{m+1} - \zeta_{m+1}^n = 0$$

Upwind method maintains unidirectional flow of information.

The Upwind Method

The inviscid Burger's equation in the following forms are unconditionally unstable:

$$\frac{\zeta_{i}^{n+1} - \zeta_{i}^{n}}{\Delta t} + u \frac{\zeta_{i+1}^{n} - \zeta_{i}^{n}}{\Delta x} = 0$$

$$\frac{\zeta_{i}^{n+1} - \zeta_{i}^{n}}{\Delta t} + u \frac{\zeta_{i+1}^{n} - \zeta_{i-1}^{n}}{2 \Delta x} = 0$$

The equations can be made stable by using backward space difference scheme if u > 0 and forward space difference scheme if u < 0 :</p>

$$\frac{\zeta_i^{n+1} - \zeta_i^n}{\Delta t} = -\frac{u \zeta_i^n - u \zeta_{i-1}^n}{\Delta x} + \text{viscous term, for } u > 0$$

$$\frac{\zeta_i^{n+1} - \zeta_i^n}{\Delta t} = -\frac{u \zeta_{i+1}^n - u \zeta_i^n}{\Delta x} + \text{viscous term, for } u < 0$$

 Upwind method of discretization is necessary in convection dominated flows.

THANKYOU