#### **Bending of beams**

A beam is defined as a rod (or) bar of uniform cross section whose length is very much greater than its other dimensions, such as breadth and thickness.

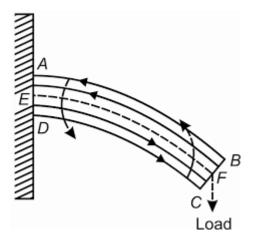
It is commonly used in the construction of bridges to support roofs of the buildings etc. Since the length of the beam is much greater than its other dimensions the shearing stresses are very small.

#### **Assumptions:**

- The length of the beam should be large compared to other dimensions.
- The applied load should be large compared to the weight of the beam.
- The cross section of the beam remains constant and hence the geometrical moment of inertia also remains constant.
- The shearing stresses are negligible.

Consider a beam ABCD, which is made up of a large number of thin plane layers are place done above the other. One end of the beam AD is fixed in the rigid support and a

Load is applied in the other end BC.



Taking the longitudinal section ABCD of the bent beam, the layers in the upper half are elongated while those in the lower half are compressed. In the middle there is a layer (MN) which is not elongated or compressed due to bending of the beam. This layer is called the *'neutral surface'* and the line (MN)at which the neutral layer intersects the plane of bending is called the *'neutral axis'*.

The layers below MN are compressed and those above MN are elongated and there will be such pairs of layers one above MN and one below MN experiencing same forces of elongation and compression due to bending and each pair forms a couple.

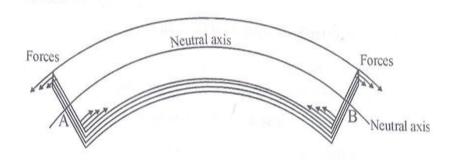
*The resultant moments of all these internal couples are called the* **internal bending moment** and in the equilibrium condition, this is equal to the external bending moment

#### INTERNALBENDINGMOMENTOFTHEBEAM

When a beam bent, the restoring couple arises. This couple balances the external couple due to external load is called **internal bending moment of the beam**. At equilibrium,

## **Restoring couple=Bending couple**

A beam may be assumed to consist of a number of parallel longitudinal metallic fibers placed one over the other and are called as filaments. Let the beam be subjected to deforming forces at its ends, due to which it bends. Let us consider a filament AB at the center of the beam. It is found that the filaments (layers) lying above AB gets elongated, while the filaments lying below AB gets compressed.

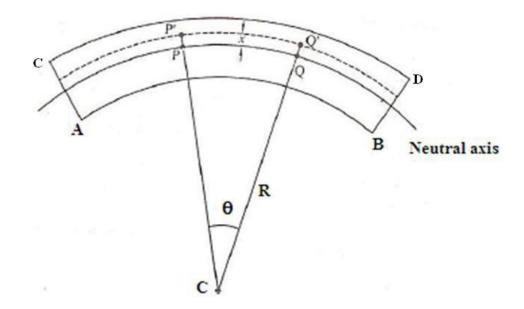


Bending of beam.





Therefore the filament i.e. layer AB which remains unaltered taken as the reference axis called as Neutral axis and the plane is called as neutral plane. Further, the deformation of any filament can be measured with reference to the neutral axis. The moment of couple due to elastic reactions (restoring couple) which balances the bending couple due to applied load is called the bending moment. Let us consider a beam under the action of deforming forces. The beam bends into a circular arc.



Let PQ be the neutral axis of the beam and P' Q' be another filament at distance y from PQ. If R is the radius of curvature of the neutral axis and $\theta$  is the angle subtended by it at its centre of curvature 'C'.

Then we can write original length **P**Q=Radius x Angle

 $= R\theta....(1)$ 

If  $R \times R$  Is the radius of curvature of the filament P'Q'.

 $P'Q' = (R x)\theta....(2)$ 

Extension Produced in the filament P'Q' due to bending=P'Q'- PQ

$$= (R x)\theta - R\theta$$
$$= x\theta.....(3)$$





Longitudinal strain=

Change in length x x Original length R R

\_\_\_\_





The Young's modulus of the filament P'Q'

Y= 
$$\frac{longitudinal stress}{longitudinal strain}$$
  
Longitudinal stress on the filament P'Q' =Y longitudinal strain  
= $y^{x}$ \_

#### R

If A is the area of cross-section of the filament, then the tensile force on the filament.

=Longitudinal stress area

$$= Y^{X} A$$
  
R

R

Moment of all the forces about the neutral axis=  $\begin{array}{c} Y \\ Ax^2 \\ R \end{array} \begin{array}{c} Y \\ I \\ R \end{array} \begin{array}{c} R^{g} \\ R \end{array}$ 

 $I_g$  is the geometrical moment of inertia and it is represented as  $AK^2$ 

Where *A*isthearea of the cross-section and *K*isradius of gyration. In equilibrium,

# Bendingmomentofthebeam=Momentofforce

Internalbendingmomentofthebeam YI(4)  $R^{g}$ 

#### **SPECIALCASES**

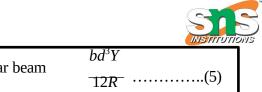
#### Rectangularcrosssection

If *b* and *d* are the bread thand thickness of the beam, then A b d and  $K_2 d^2 \frac{1}{12}$ 

Using the value of  $I_g$ 

 $\frac{bd^3}{12}$  inequation(4)





□Internalbendingmomentoftherectangular beam





## Circularcrosssection

If r bethere a discrete the formula of the beam, then  $A r^2$  and  $K_2 r^2 - \frac{1}{4}$ 

 $I_g AK \stackrel{2}{\phantom{a}} r \stackrel{4}{\underline{\phantom{a}}}$ 

Usingthevalue

 $I_g = \frac{r^4}{4}$  inequation(4)

Ter A sum all and the sum and a failer attended a sum	r4Y	
Internalbendingmomentofthecircularbeam	-4R	(6)

#### DEPRESSIONOFACANTILEVER

#### DEFINITION

A light beamclamped horizontally at one end and loaded with a weight W = Mg at the free end is called a **cantilever**. Inequilibrium,

Externalbendingmoment =Internalbendingmoment

#### THEORY

Let us considerabeamfixed atone endand loaded at its other freeend as shown in fig1.7.2.2.Let AB is the neutral axis of a cantilever (a beam or rod) of length 'l' is fixed at the end A and loaded at the free end B by a weight W. Due to load applied the cantilever is depressed to B'.

LetBB'represents the vertical depression at the free end.





Due to the load applied at the free end, a couple is created between the two forces. (i.e)

(i) Force(load'W')appliedatthefreeendtowardsdownwarddirectionand





#### (ii) Reaction(R)actinginthe upwarddirectionatthesupportingend.

This external bending couple tends to bend the beam in the clockwise direction. But, since one end of the beam is fixed, the beam cannot rotate. Therefore the external bending couple must be balanced by another equal and opposite couple, created due to the elastic nature of the body called as internal bending moment.

Consider the section of the cantilever P at a distance 'x' from the fixed end A.Q is another point at a distance dx from P i.e., PQ = dx. It is at a distance (l-x) from the loaded end B'. Considering the equilibrium of the portion PB', there is a force of reaction W of P. Let O be the centre of curvature and R be the radius of curvature.

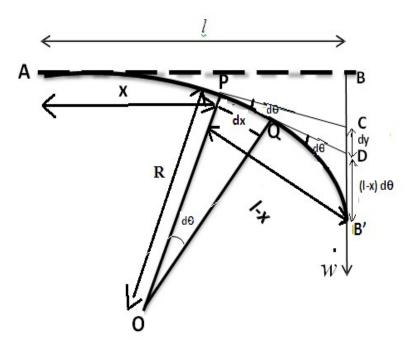
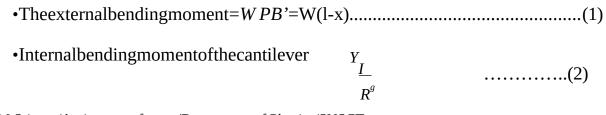


Fig1.7.2.2.Depressionofacantilever.





R-RadiusofthecurvatureoftheneutralaxisatP.

WhereY–Young'smodulusofthecantilever.

 $I_g$ -Geometricalmomentofinertiaofitscross-section. In the equilibrium position,

External bending moment=Internal bending moment  $W(l x) \stackrel{Y}{I}$  \_\_\_\_\_  $R^{g}$  \_\_\_\_\_  $R \stackrel{YI_{g}}{}$  ......(3)

Let Q be another point on the bent cantilever at small distance 'dx' from P. Since P and Q are very near, we can assume that the radius of curvature R is practically the same.

The tangents are drawn at P and Q meeting the vertical line BB' at C and D. Let  $d\theta$  be the angle between the tangents at P and Q.

Sin  $d d R^{dx}$  \_\_\_\_\_  $R^{dx} dx$  \_\_\_\_\_ Substituting the value of R from(3) in(4), we have dx

$$\frac{dX}{d} = \frac{YI_g}{W(lx)}$$

 $d_{YI_g}^{W(l \ x)dx}$ 

If 'dy' is the depression due to the curvature at PQ dy (l x)d

.....(6)

.....(5)

Substituting value of  $d\theta$ 

$$dy (l x) \frac{W(l x)}{M_g} dx$$

$$dy \frac{W(l x)^2}{M_g} dx$$

$$(7)$$

M.Sriram /Assistant professor /Department of Physics/SNSCT



# R-RadiusofthecurvatureoftheneutralaxisatP.



To find the total depression at the free end of the cantilever equation (7) has to be integrated

from 0 to l.

dy W

$$\int_{0}^{l} \frac{(l x)^{2}}{YI_{q}} dx$$





12

Depressiony 4Mgl3 Ybd3

.....(9)

# Circularcrosssection

If the theradius of the beam, then  $A r^2$  and  $K^2 \frac{r^2}{2}$ Substituting the values of  $I_g AK^2 r^4$  and W Mg in equation (8) Depressiony  $\frac{4Mgl3}{3Yr4}$  .....(10)

# EXPERIMENTAL DETERMINATION OF YOUNG'S MODULUS BY CANTILEVER DEPRESSION METHOD

# CONSTRUCTION

M.Sriram /Assistant professor /Department of Physics/SNSCT





One end of the beam is rigidly clamped at one end to the edge of the table using G-clamp. A tall pin P is fixed vertically to the free end of the bar. A loop ofcottonstringorahookisattachedtothisendofthebarandaweighthangeris





suspended from it. A travelling microscope is focused on the tip of the pin as shown in fig.

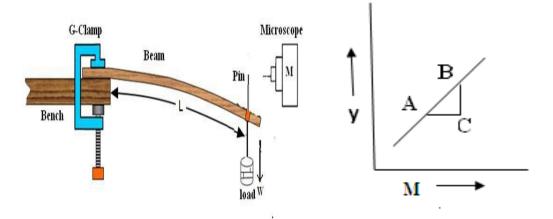


Fig1.7.2.3Experientalverificationofacantilever.

#### PROCEDURE

A dead load without any slotted weights is attached to the hook. The microscope is adjusted such that the horizontal cross wire coincides with the tip of the image of the pinand the readingonthe verticalscale is taken. Loads are added to the hanger insteps of 50g and every time, the readings are noted on the vertical scale. A travelling microscope is focused on the tip of the pin as shown in fig. These observations are also repeated while unloading in the same. Steps and the readings are tabulated. The mean depression 'y' for a load 'M' kg is found from the tabulated readings. Theobservations are tabulated as follows.

#### Graphicalmethod

A graph is drawn between the load (M) along X-axis and elevation (y) along Yaxis. It is found to be a straight line as shown in fig. The slope of the straight line gives the value of (y/M).





	Microscopereadings							depression	Mean
	Loading			Unloading			Mean	(y)for100g	(M/y)
Load	MSRc	VSR	TR	MSRc	VSR	TR	-	10 <sup>2</sup> m	
10 <sup>3</sup> kg	m	cm	cm	m	Cm	Cm	10 <sup>2</sup> m		Kgm <sup>-1</sup>
W									
W+50									
W+100									
W+150									
W+200									

Hence Young's modulus of the cantilever can be calculated as

- $4gl^3 M$ 
  - $bd^3$

y