

SNS COLLEGE OF TECHNOLOGY

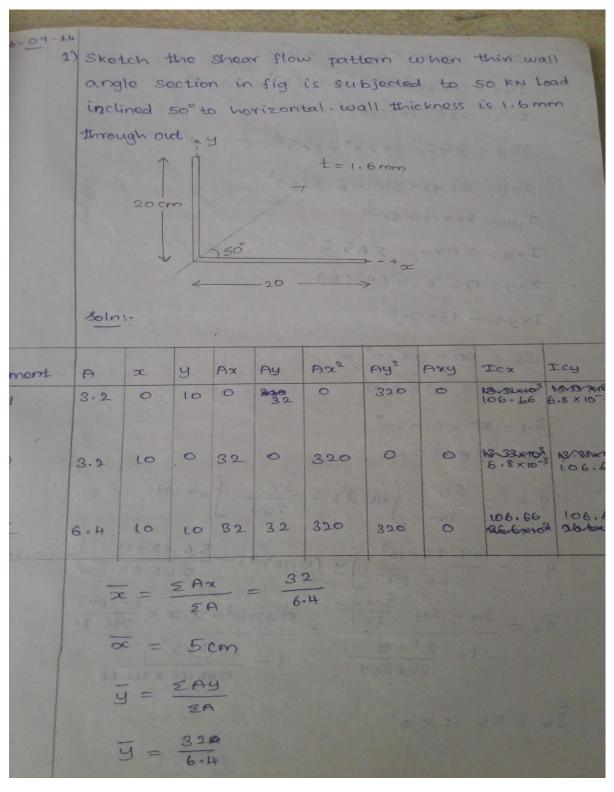


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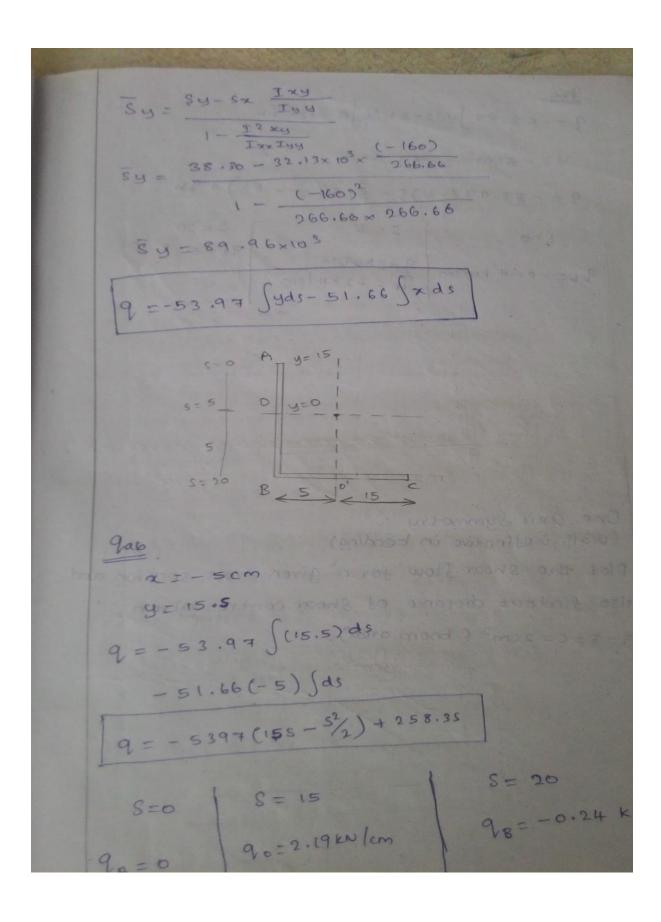
DEPARTMENT OF AEROSPACE ENGINEERING

Subject Code & Name: 19AST203 Aircraft Structural Mechanics

TOPIC: Shear flow in open sections with one axis of symmetry



Txy = - 160 cm 4 Sx = 50 x cos 50° \$ Sx = 32.13×103N Sy = 50 x 8in 50 Sy = 38.30 N 9 = - Sy geds - Sx Jxt ds $9 = -\frac{89.96 \times 10^3}{266.67} \int y (0.16) ds - \frac{86.12 \times 10^3}{266.67} \int 9$ $\overline{S}_{x} = \frac{S_{x} - S_{y}}{1 - \frac{I^{2} \times 4y}{I_{xy}}} = \frac{82.13 \times 10^{8} - 38.30 \times \frac{(-160)}{266.66}}{1 - \frac{I^{2} \times 4y}{I_{xx} I_{yy}}} - \frac{(-160)^{2}}{1 - \frac{(-160)^{2}}{2}}$



53.97 Jyds-51.66 Jads + 96 07-14 One axis Symmetry (wall ineffective in bending) 1) Plot the shear flow for a given cross secti also findout distance of Shear centre giver A = B = C = 2 cm2 (boom area).