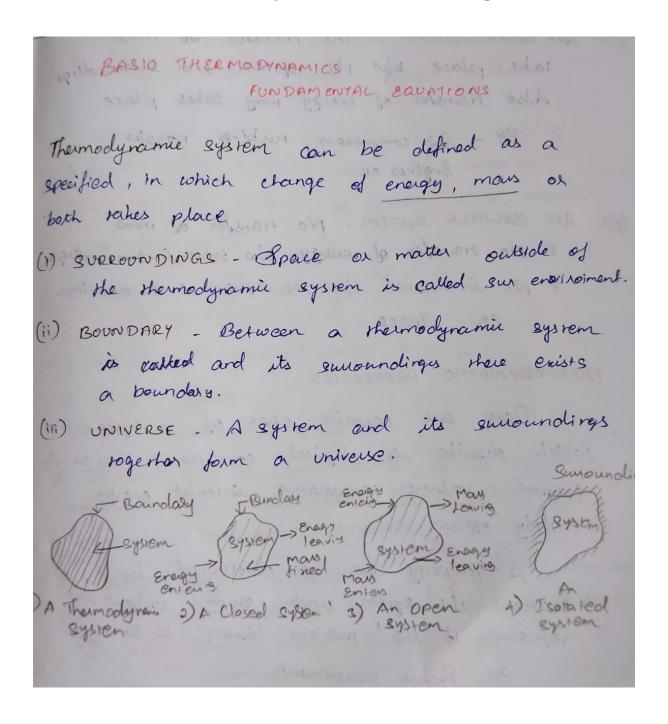


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DEPARTMENT OF AEROSPACE ENGINEERING 19ASB204 - Aerospace Propulsion Basic Thermodynamics Fundamental Equation





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Three Kinds of Thermodynamic dystems

- (a) closed system (b) onen system (c) Isolaved system
- (a) A CLOSED SYSTEM It is a system which the mass remains constant. No transfer of mass across the system boundary.

There may be transfer of energy blw system and its sunoundings.

Eg - Steam Engine. Pisson - Cylindes.

- (b) AN open system. The transfer of mass take place blue the system and sunoundings Also transfer of energy may takes place.
 - Eg Air commesons, rubines, noggles Engines ex.
- (c) AN SSOLATED SYSTEM NO transfer of mous

 * No transfer of energy blw system & sumoundings

 * No interaction blw the system & sumoundings.

 Eg Space

THERMODYNAMIC PROPERTIES

There are certain char of a system which describe its physical conditions. They are Pressul, volume, remperature, internal energy, enthaly ex.

A property can be elevermined by examine the examine under order order and



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Propersies are grouped into 2 caregories a) Intensive Prop

b) Entensive Prop a) Intensive - Prop which are independent of the mans of the system eg - Pr. Temp, Spv, sp In Eno 3p Entropy. b) Extensive - Prop which are depend upon the mas of the system. The magnitude of an extensive moperty depends on mais of system Eg - Volume, Internal Energy, Enthalpy, The reisio of an ext prop to the mais of a system called specific value of morrer my. eg. It rord volume of system is v. mans.m relific valume
(or)

Volume per Vrit mens

Of the system My , specific internal energy u = m * Caps - Extensive * Small - specific values of non I Internal Prop. Charecreissics of a system within the equilibrium system. They are - mans, composition, remperature, internal energy etc. * Enternal Prop - Charecrevisitis of either the motion on the position of the system in a field of force. They are leineric energy, Porential energy,







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31A1E, PATH, PROCESS & CYCLE

(i) The moperies of a system can be used as coordinates to describe the condition of the system. This condition is known as state of system.

(ii) Path - The succession of state passed through during a charge of state forms the path of the charge of state.

$$mC_{V}(\tau_{1}-\tau_{2}) = \frac{mR(\tau_{1}-\tau_{2})}{\gamma-1}$$

$$C_{V}(\tau_{1}-\tau_{2}) = \frac{R(\tau_{1}-\tau_{2})}{\gamma-1}$$