

Unit-3 2D problem

Strain-Displacement [B] matrix [Gradient matrix]

HKT

Strain in x-direction

$$\epsilon_x = \frac{\partial U}{\partial x} = \frac{\partial N_1}{\partial x} \cdot u_1 + \frac{\partial N_2}{\partial x} u_2 + \frac{\partial N_3}{\partial x} u_3$$

Strain in y-direction

$$\epsilon_y = \frac{\partial V}{\partial y} = \frac{\partial N_1}{\partial y} \cdot v_1 + \frac{\partial N_2}{\partial y} v_2 + \frac{\partial N_3}{\partial y} v_3$$

Strain along xy direction.

$$\gamma_{xy} = \frac{\partial U}{\partial y} + \frac{\partial V}{\partial x} = \frac{\partial N_1}{\partial y} \cdot u_1 + \frac{\partial N_2}{\partial y} u_2 + \frac{\partial N_3}{\partial y} u_3 + \frac{\partial N_1}{\partial x} \cdot v_1 + \frac{\partial N_2}{\partial x} v_2 + \frac{\partial N_3}{\partial x} v_3$$

In matrix form

$$\begin{Bmatrix} \epsilon_x \\ \epsilon_y \\ \gamma_{xy} \end{Bmatrix} = \begin{bmatrix} \frac{\partial N_1}{\partial x} & 0 & \frac{\partial N_2}{\partial x} & 0 & \frac{\partial N_3}{\partial x} & 0 \\ 0 & \frac{\partial N_1}{\partial y} & 0 & \frac{\partial N_2}{\partial y} & 0 & \frac{\partial N_3}{\partial y} \\ \frac{\partial N_1}{\partial y} & \frac{\partial N_1}{\partial x} & \frac{\partial N_2}{\partial y} & \frac{\partial N_2}{\partial x} & \frac{\partial N_3}{\partial y} & \frac{\partial N_3}{\partial x} \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \\ v_1 \\ v_2 \\ v_3 \end{Bmatrix}$$

$$N_1 = \frac{P_1 + q_1 x + r_1 y}{2A}; \quad N_2 = \frac{P_2 + q_2 x + r_2 y}{2A}; \quad N_3 = \frac{P_3 + q_3 x + r_3 y}{2A}$$

$$\frac{\partial N_1}{\partial x} = \frac{q_1}{2A}; \quad \frac{\partial N_2}{\partial x} = \frac{q_2}{2A}; \quad \frac{\partial N_3}{\partial x} = \frac{q_3}{2A}$$

$$\frac{\partial N_1}{\partial y} = \frac{r_1}{2A}; \quad \frac{\partial N_2}{\partial y} = \frac{r_2}{2A}; \quad \frac{\partial N_3}{\partial y} = \frac{r_3}{2A}$$

$$\{\epsilon\} = \begin{Bmatrix} \epsilon_{xx} \\ \epsilon_{yy} \\ \gamma_{xy} \end{Bmatrix} = \frac{1}{2A} \begin{bmatrix} q_1 & 0 & q_2 & 0 & q_3 & 0 \\ 0 & r_1 & 0 & r_2 & 0 & r_3 \\ r_1 & q_1 & r_2 & q_2 & r_3 & q_3 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \\ v_1 \\ v_2 \\ v_3 \end{Bmatrix}$$

This is of the form

$$\{\epsilon\} = [B] \{U\}$$

Stress-Displacement Relation

$$\{\sigma\} = [D] \cdot \{\epsilon\}$$

$$\{\sigma\} = [D] \cdot [B] \cdot \{U\}$$

Max. Normal Stress

$$\sigma_1 = \frac{1}{2} \left[(\sigma_x + \sigma_y) + \sqrt{(\sigma_x - \sigma_y)^2 + 4 \tau_{xy}^2} \right]$$

$$\sigma_2 = \frac{1}{2} \left[(\sigma_x + \sigma_y) - \sqrt{(\sigma_x - \sigma_y)^2 + 4 \tau_{xy}^2} \right]$$

principle angle $\theta_p = \frac{1}{2} \tan^{-1} \left[\frac{2 \tau_{xy}}{\sigma_x - \sigma_y} \right]$

WKT stiffness matrix $[K]$ for CST

$$[K] = \int [B]^T [D] [B] \cdot dV$$

$$= [B]^T [D] [B] \cdot A \cdot t$$

where t = Thickness of body is case of plane stress

= 1 in case of plane strain